



Control and
communication
architecture of the
GLORIA limb imager

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This discussion paper is/has been under review for the journal Atmospheric Measurement Techniques (AMT). Please refer to the corresponding final paper in AMT if available.

In-flight control and communication architecture of the GLORIA imaging limb-sounder on atmospheric research aircraft

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Received: 30 October 2014 – Accepted: 14 January 2015 – Published: 10 February 2015

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Published by Copernicus Publications on behalf of the European Geosciences Union.

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Abstract

The Gimballed Limb Observer for Radiance Imaging of the Atmosphere (GLORIA), a Fourier transform spectrometer based limb spectral imager, operates on high-altitude research aircraft to study the transit region between the troposphere and the stratosphere. It is one of the most sophisticated systems to be flown on research aircraft in Europe, requiring constant monitoring and human intervention in addition to an automation system. To ensure proper functionality and interoperability on multiple platforms, a flexible control and communication system was laid out. The architectures of the communication system as well as the protocols used are reviewed. The integration of this architecture in the automation process as well as the scientific campaign flight application context are discussed.

1 Introduction

The Gimballed Limb Observer for Radiance Imaging of the Atmosphere (GLORIA) is an infrared limb sounder which is based on an imaging Fourier transform spectrometer. It is designed for atmospheric research in the upper troposphere and lower stratosphere (UTLS) region (Riese et al., 2014; Friedl-Vallon et al., 2014). The instrument can operate on various airborne platforms: as of yet, it has flown on the German high altitude research aircraft HALO and on the Russian stratospheric aircraft M55 Geophysica. The instrument was successfully deployed on three measurement campaigns: during the combined TACTS – standing for Transport and Composition in the Upper Troposphere and lower-middle Stratosphere – and ESMVal – standing for Earth System Model Validation – campaigns in 2012 (Engel and Boenisch, 2013; Oelhaf et al., 2013) as well as during the ESSenCe – standing for ESa Sounder Campaign – campaign in 2011 (Woiwode et al., 2014, and references therein).

The instrument is mostly designed to sound the atmosphere in across-track limb geometry, but can also proceed with measurements in nadir geometry. In addition, to

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2 Instrument overview

The 48 kg spectrometer is mounted in a gimballed frame assuring both line of sight orientation and stabilisation as well as compensation of platform movements. This construction ensures a sufficient pointing accuracy and stability during measurement to achieve the required instrument performances (Friedl-Vallon et al., 2014). This system, being installed outside of the aircraft pressurised cabin in a belly pod or partially exposed instrument bay, must operate at temperatures ranging from 200 to 320 K and pressures ranging from sea level pressure to 70 hPa, which adds to its inherent complexity and leads to a need for constant monitoring (Piesch et al., 2014).

Along the science data delivered by the spectrometer, more than a thousand housekeeping parameters of the instrument and its subsystems are acquired during operation. This information is used in real time for monitoring of the instrument to ensure proper function. The data is also stored for detailed post-flight analysis. Analysis and results of some of these housekeeping data is described by Piesch et al. (2014). The housekeeping data is also used to support processing of the science data as it also includes geolocation and pointing information. Additional data is provided by environmental sensors such as temperature and vibration sensors, as well as diagnostic information such as the voltages and currents that are provided to the different systems.

The full housekeeping data rate is typically around 110 kB s^{-1} , depending on which sub-units are active and sending data. This data rate is significantly smaller than the science data rate produced by the spectrometer, which typically reaches 74 MB s^{-1} (or 260 GB h^{-1}). The science data is transferred to an on-board storage over a dedicated optical link using the Camera Link protocol. This link is completely independent of the control and communication architecture presented in this paper. Since it is too large to be processed, transmitted to ground and reviewed in real time, the housekeeping data also includes diagnostic information on the science data and on its overall quality.

To support a large number of sensors placed on different levels of the gimbal frame, a modular approach has been chosen. The data acquisition and control tasks have

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been separated into different interdependent units. Each unit sends its housekeeping data flow to, and receives commands from a Central Control Computer. The unit intercommunication and communication with the Central Control Computer is based on a Gigabit Ethernet (GbE) Local Area Network (LAN). Address assignment is done via the Dynamic Host Configuration Protocol (DHCP) by the Central Control Computer and any number of units may be integrated in the data handling concept.

The Central Control Computer receives and stores all housekeeping data as well as the science data from the spectrometer and provides a gateway to external networks for control and monitoring. Notably, it relays to the electronic units the commands issued from the operators, which are either on board or at various ground stations. Furthermore, the Central Control Computer plays a key role in automation of the system by running the automatic control software.

The modularity of the communication infrastructure, notably the physical separation between the Central Control Computer and the instrument, is an important element in its portability between platforms with different requirements and facilities. On the HALO aircraft, the instrument itself is mounted in the belly pod of the aircraft while the Central Control Computer is mounted in an instrument rack in the aircraft cabin (Friedl-Vallon et al., 2014; Giez, 2010). An operator can be present on board. On the M55 Geophysica, which has no pressurised cabin, the whole instrument including the Central Control Computer is mounted in an instrument bay in the nose (Friedl-Vallon et al., 2014). No operator can fly along the instrument on this single-seat aircraft.

Despite the instrument complexity both in its design and operation, GLORIA can be fully remotely operated by a ground crew, mitigating the need for an operator on board the aircraft and enabling operation on aircraft such as the M55 or future deployment on balloon-borne platform or unmanned aircraft systems. An operator accompanying the instrument is nevertheless required if the data link with the aircraft does not meet the stability and throughput requirements for remote operation, control and monitoring. An on-board operator may also be required for campaign-specific needs, as later discussed in Sect. 7.

3 Network architecture

Primarily for practical reasons, but also for performance and security reasons, the network topology used for the GLORIA sub-systems is divided in two distinct physical networks: the GLORIA-LAN (Glo-LAN) interconnects all the control and housekeeping units of the instrument with the Central Control Computer while the Laboratory-LAN (Lab-LAN) interconnects all control and monitoring stations on ground. This topology is illustrated in Fig. 1.

The data transfers between the Glo-LAN and the Lab-LAN are done over a single entry point, the so-called Bridge Computer, using a third dedicated interfacing network. Except for the Bridge no other ground-based control or monitoring computer has a direct connection to the on-board Central Control Computer.

Depending on the aircraft and the available infrastructure, the third network may be routed over a narrowband one-channel Iridium satellite link, over a broadband multi-channel Aero Openport Iridium link or over a broadband Inmarsat link. Furthermore, during on-ground operations, the third network may be routed over Wireless LAN (WLAN). The M55 aircraft has no communication facilities for the payload and thus the Iridium Aero Openport satellite communication system has been installed as part of the GLORIA experiment. The HALO aircraft offers a broadband Inmarsat system and the addition of an Iridium system similar to the one flown on the M55 is under investigation. The combination of Inmarsat and Iridium is already available on the HIAPER, a similar platform operated by NSF/NCAR (Laursen et al., 2006). The nature of the bridging network and its exact topology can be changed at any time. These changes remain transparent to the users.

Table 1 details the available bandwidth over the different bridging networks.

For measurement campaigns which require one or more remote operation bases, for example for overnight stay and refuelling, a fourth network is deployed at the remote site. This network allows connection of the operators at the remote site with the Lab-LAN. The connection is established securely through a virtual private network (VPN)

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over the locally available Internet infrastructure (e.g. public WLAN or GSM/UMTS). The Central Control Computer of GLORIA can also connect to this remote site network over WLAN for faster data transfer and to limit the costs of operating satellite links while on ground.

The VPN used for the remote sites is based on the OpenVPN protocol and allows any computer on the Internet with the proper certificates to securely join the control and monitoring network of GLORIA, the Lab-LAN. This allows technical operators to monitor and command the instrument without having to be on site. In addition, a second VPN is established between the Bridge Computer and the Central Control Computer. This allows secure and transparent bi-directional communication through the infrastructure and firewalls of the satellite communication service providers, which otherwise block up-going communications. The use of VPN is illustrated in Fig. 2.

Additional networks dedicated to specific tasks are available within the aircraft cabin on HALO. A service port at the Central Control Computer allows the connection of the on-board operator directly into the instrument Glo-LAN, thus allowing the operator to communicate directly with the control and housekeeping units of GLORIA for low-level access. A transfer network dedicated to transfer the science data from the Central Control Computer to a storage unit during the on-ground maintenance is also available. A third dedicated cabin network connects the Central Control Computer to the public cabin network of HALO, giving access to avionic data and condition data provided by standard HALO sensors (Giez, 2010; Krautstrunk and Giez, 2012). The public cabin network also gives access to a GPS-based time server. The multiple networks are managed through virtual LAN (VLAN) following the IEEE 802.1Q standard (IEEE, 2011) with the help of a managed switch mounted in the instrument rack alongside the Central Control Computer.

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4 Housekeeping data handling

The housekeeping data transmission is based on a protocol which evolved from one previously used with GLORIA predecessor instruments, MIPAS-B2 (Friedl-Vallon et al., 2004) and MIPAS-STR (Piesch et al., 1996). The use of the same data structure, which was perfectly adapted for links with limited bandwidth, allowed in large part to reuse the existing logical architecture and implementation and ensured backward compatibility. The former serial link frame protocol, including the sync word and header, was simply encapsulated in Transfer Control Protocol (TCP) frames.

Each unit of GLORIA opens a TCP connection on a dedicated port with a server application called *Spider* running on the Central Control Computer. Following the connection establishment, each housekeeping data packet is transmitted with one or, when the frame is bigger than the maximum transmission unit (MTU), with multiple TCP frames. Each packet is then written as received by the Spider application to a logging file on disc. The sync word inherited from the serial protocol found in the data packet header facilitates the interpretation of the raw packet log, where packets may have different length and formats.

The structure of a housekeeping data packet, illustrated in Fig. 3, consists of a data block preceded by a header, starting with the constant sync word. The header also includes the unique identifiers of the sending device (DEV) and of the data packet type (TAG), the packet type serial counter, the packet send time, the packet length and a cyclic redundancy check (CRC) value for error detection.

A device may send only a single type of data packet or different types of data packets, each identified with a TAG. The data structure of each type of data packet for each unit is predefined in a definition library used by the different programs to interpret the data. The library specifies the name of each data entry, the data type, size, physical units and, if applicable, offset and scaling factors. Using this definition library, any program can interpret the raw data packet for any use such as processing of the data or visualisation.

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The data packets may be sent by units at a constant rate or sporadically, triggered by events. Each packet type, even those coming from the same unit, may be sent at different rates to accommodate the needs related to the housekeeping values they convey. For housekeeping parameters produced at high frequency, for example vibration and acoustic data, the defined data structure may repeat itself multiple times within a single data packet up to the length given in the packet header. These so-called cyclic data packets allow the transmission of a large multiple of measured parameters within a single block, reducing the number of required packets, thus reducing the communication overhead on high rate data.

The implementation of the communication protocols in embedded controllers of electronic units is standardised through a universal C library. This library has been effectively used on multiple platforms, including Windows, Linux and QNX. Furthermore, the communication protocols have been readily implemented in Java and Groovy programs and are available as modules for Labview programs. The broad support of different platforms gives a large flexibility in the design of the different sub-units of GLORIA, as well as a lot of liberties during prototyping and testing of these units.

4.1 Centralised data collection and distribution

Every unit, or device, sends its data packets to the Spider server application located on the Central Control Computer. Not only logging the received data, the Spider application can also redistribute it, or parts of it, to other computers for visualisation or further retransmission. Due to the nature of the data link between the ground station and the aircraft, it is neither convenient nor desirable for single ground operators to receive housekeeping data destined for visualisation directly from the Central Control Computer.

For this task, a second Spider server application runs on the Bridge Computer. This second Spider builds a connection with the Spider on the aircraft and receives, logs and eventually retransmits every data packet sent by the in-flight server to the users requesting them. The connection can either be built through a TCP channel when all

4.2 Data visualisation

The data visualisation tool receives the data packets exactly as sent by the originating unit. It is only at the end level of the whole data collection chain that the data packets are interpreted. The interpretation is done using the parameters specified in the definition library for each data packet type. The definition library is laid down as a simple text format file and is interpreted at runtime by the visualisation tools. No update of the software tools is required as the data packet structure evolves or as the format of the values therein are changed.

Within the visualisation tool, the operator selects the housekeeping parameters he desires to monitor. The tool then requests the relevant data packet from the Bridge Spider application over a TCP connection with a special filter request data packet. Following the request, the Spider retransmits the requested data packets to the visualisation tool over the existing TCP link. This process is illustrated in Fig. 4.

Figure 5 shows a typical housekeeping visualisation window. In addition to raw values, selected parameters can also be displayed as graphs. The visualisation tool shows the name of the data values, the values themselves after proper scaling and offset correction as well as the physical units. A green indicator shows whether the values are actively updated or if the values displayed are old. Warning flags on the far right can indicate if a value is within its nominal range, which is also defined within the definition library.

As deferred review and visualisation of the housekeeping data may be convenient, the housekeeping feed incoming at the Bridge is also sorted into a database. This database contains the housekeeping parameters in their interpreted form based on the structure defined in the definition library file. The values within the database can be reviewed and visualized with time-delay, making it a great asset when, for example, reviewing the events that led to an issue which occurred in flight.

In addition to the technical housekeeping visualisation tool, general information about the missions are published live on a website. The web application requests the desired

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housekeeping data from the Bridge computer and publishes it on the Internet, presenting information about the instrument status as well as the aircraft altitude and location. This simple visualisation tool allows the large scientific community working with GLORIA, but also with other instruments on board the research aircraft and moreover the broader scientific community interested in specific HALO missions, to follow actively the progress of the missions.

5 Control interface

Similar to the housekeeping protocol, the basis for the control protocol also dates back to the communication used for the predecessor instruments of GLORIA, MIPAS-STR and MIPAS-B2. As with the housekeeping data packet, the control data packets used to issue instructions to a specific unit have been encapsulated in TCP/IP frames for the transmission over the modern infrastructure of GLORIA.

A control data packet – or telecommand packet (TC) – consists of 10 bytes, starting with a synchronisation byte. The packet, detailed in Fig. 6, includes identifier (ID) of the unit to be commanded, ID of the given command plus 4 bytes optional values, the type of the optional values, a parity check value and the ID of the command originator. The type flag indicates to the receiving unit command interpreter how the 4 value bytes should be interpreted. Table 2 presents how the different optional value bytes may be used according to the type parameter.

A shell interface interpreting human readable commands is available to the operators. Each time a command is entered, the shell opens a new TCP connection with the Spider server application and sends the interpreted commands as a 10 bytes payload. The on-board Spider determines the proper destination device address using the information contained in the payload and redirects the command to that specific unit in the GLORIA-LAN. Upon successful retransmission, the payload is sent back as acknowledgement (ACK) in reverse byte order to the originator by the Spider application over the same link. This validation process is illustrated in Fig. 7.

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and of the TACTS and ESMVal campaigns in 2012. The flexibility of the communication systems allows operation on different platforms with varying physical infrastructure. Also, the embedded Ethernet technology used for the instrument internal network has proved to be a wise and flexible choice. Units can be changed, upgraded or added with little or no modification to the wiring concept. Changes in the infrastructure, even during operation, are transparent to the instrument operators.

Visualisation tools for the technical housekeeping data allow operators both on board and on the ground to monitor the operation of the instrument. A flexible low-bandwidth protocol was established to cope with any type of available data links between the airborne platform and the ground stations. The transmitted data may be changed ad hoc and according to the available bandwidth. General visualisation of selected data over web interface is also available for a broader community.

The infrastructure supports decentralised operation centres, all communicating with the instrument through a single point of entry. Remote operators connected to the Internet can join the control and monitoring network over secure VPN connections. This possibility simplifies the personal resource management and reduces the burden on specialised technical staff for long duration and decentralised campaigns.

Both the infrastructure and protocols deployed for the GLORIA experiment are flexible and scalable such that they could be deployed with little modifications for any experiment. Even multiple experiments might share a common infrastructure. Commanding of instruments and devices over serial link is also thinkable through the use of Ethernet-Serial bridges. This would be facilitated by the form of the payload frame used, which is inherently compatible with serial data links.

Complex instruments like GLORIA significantly rely on automation systems which were successfully implemented. The scripting based automation system uses the standard communication infrastructure to gain information on the instrument status and to control it. Many operations typically executed by on-board operators during flight on various scientific payloads could be handled by proper scripting, mitigating the need for operators and thus optimising scientific payload capacity.

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Table 1. Available networks for the connection between the Central Control Computer on the aircraft and the Laboratory-LAN. No nominal bandwidth is given for the Inmarsat system as it was not used as of yet. The nominal bandwidth is the bandwidth typically used for commanding and for the housekeeping feed. Extraordinary activities such as running a remote desktop connection between the ground and the Central Control Computer are not included in this nominal bandwidth.

Bridging network	Specified Bandwidth	Nominal Bandwidth	Availability
Single-Channel Iridium	2.4 kbit s ⁻¹	1 kbit s ⁻¹	HALO
Iridium Aero OpenPort	150 kbit s ⁻¹	25 kbit s ⁻¹	M55, HALO*
Inmarsat Swift 64	64 kbit s ⁻¹	–	HALO
Wireless LAN	54 Mbit s ⁻¹	440 kbit s ⁻¹	M55, HALO

* Under consideration and investigation for future implementation at time of publishing.

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Table 2. Usage of the control data packet optional values according to the specified type. If relevant, the most significant byte (MSB) and the least significant byte (LSB) are indicated. The little-endian convention is followed.

Type	Description	Value Usage			
		0	1	2	3
0 × 00	2 × char	char 1	char 2	<i>unused</i>	<i>unused</i>
0 × 01	short	MSB	LSB	<i>unused</i>	<i>unused</i>
0 × 02	long	MSB	byte 2	byte 3	LSB
0 × 03	float (IEEE 754)	MSB	byte 2	byte 3	LSB
0 × 04	2 × short	MSB 1	LSB 1	MSB 2	LSB 2

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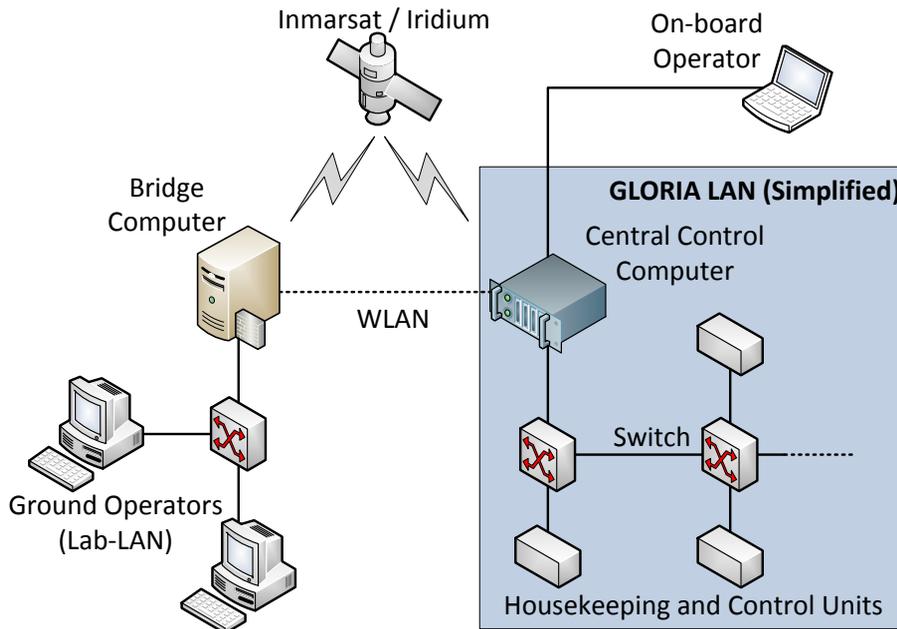


Figure 1. Overview of the network topology. On the left side, the ground infrastructure communicates with the on-board infrastructure, shown on the right side, through satellite up and down link or through WLAN.

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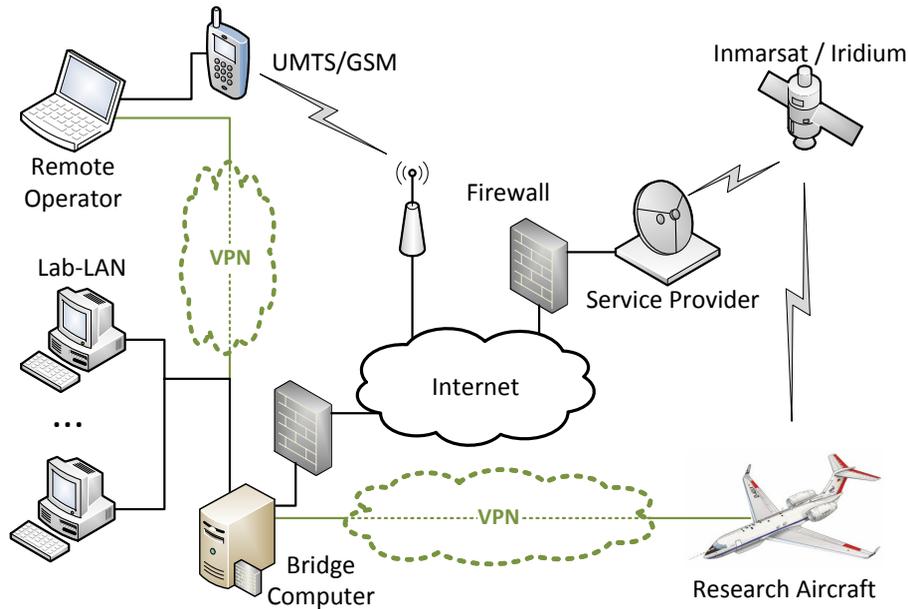


Figure 2. Global network topology diagram showing how the data transmission is encapsulated through VPN connection (shown in green) while transiting through the physical networks (shown in black). Here the logical (VPN) and physical networks are illustrated in parallel to highlight how those are perceived by the users. The operators at remote sites are fully integrated in the Lab-LAN through the VPN connection.

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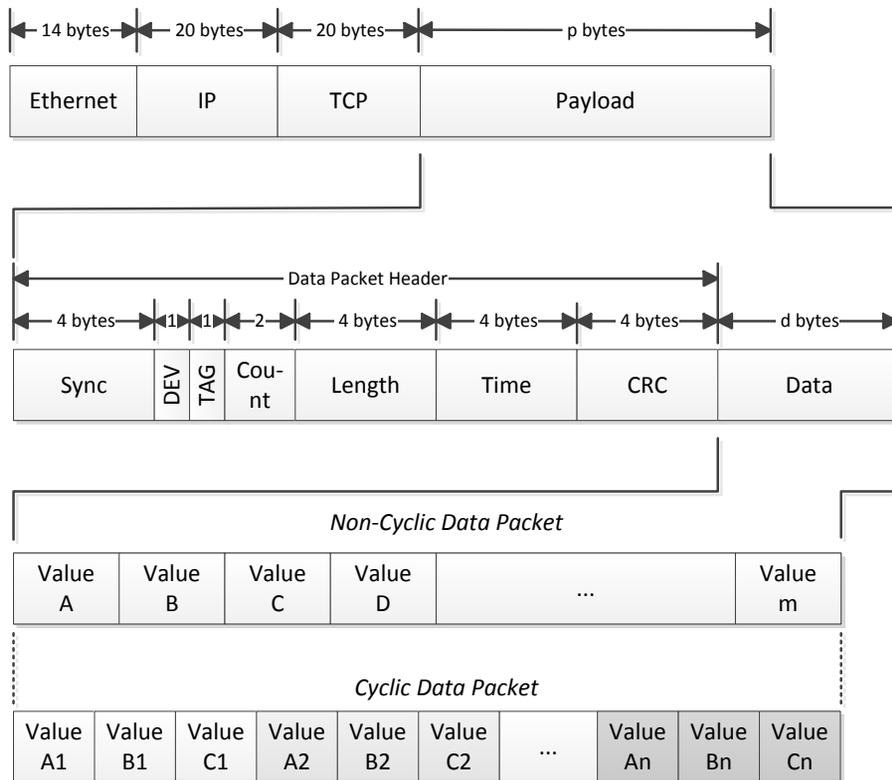


Figure 3. Structure of a GLORIA housekeeping data packet, with its header, encapsulated in a TCP/IP frame. The data block of the data packet can be either non-cyclic where every defined data value is only included once, or cyclic where each defined value is repeated n times. A mixed mode where the first values are present only once and the last ones are repeated as with the cyclic mode is also possible (not illustrated). The length of the data packet includes the header as well as the n repetitions of a cyclic data packet.

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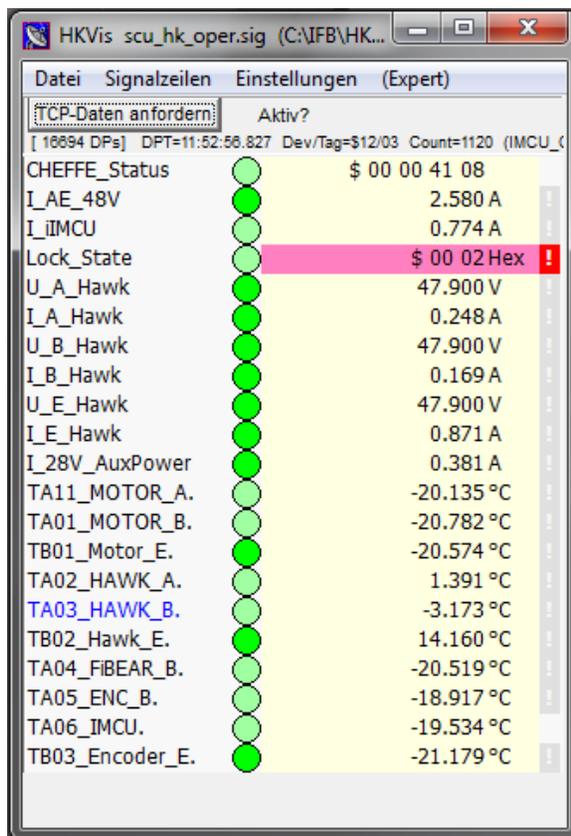


Figure 5. Typical housekeeping visualisation showing parameters transmitted over different data packets. The disposition of the values as well as the displayed information such as units or out of range indicators can be chosen as desired by the operator.

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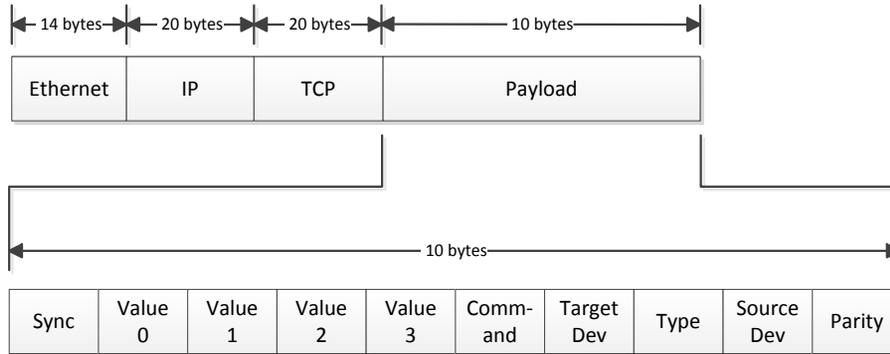


Figure 6. Structure of a GLORIA telecommand data packet encapsulated in a TCP/IP frame. The synchronisation byte is followed by four optional value bytes and by the command byte, indicating which action to take. In addition, the target device as well as the source device are included in the packet. The Type byte indicates how the 4 value bytes should be interpreted as shown in Table 2.

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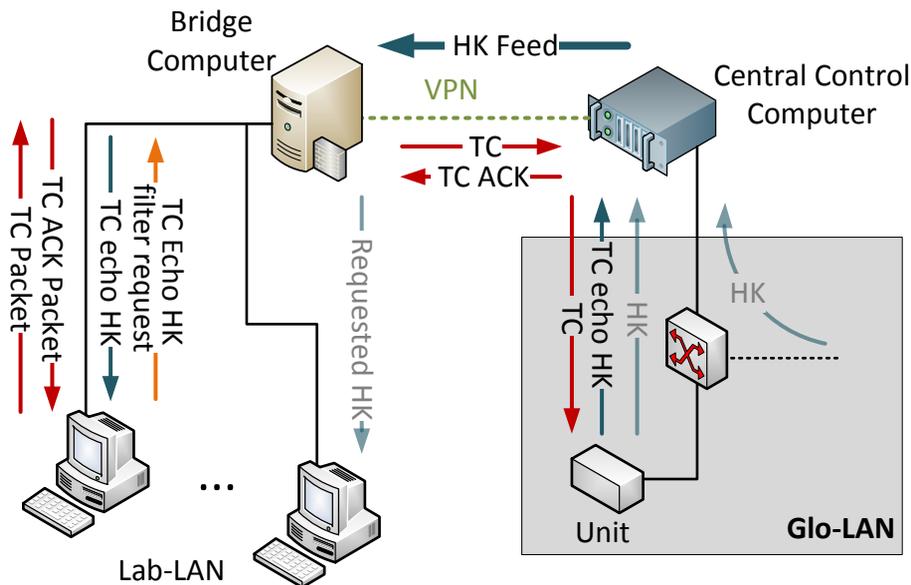


Figure 7. Telecommand (TC) and Housekeeping (HK) relaying between the operators and the GLORIA Housekeeping and Control Units. The relaying takes place over the cascaded Spider Application running on the Central Control Computer and on the Bridge Computer. The TC acknowledge (ACK) is sent back from the Spider on the Central Control Computer to the originating operator's shell upon delivery of the TC packet to the unit, which also sends an echo confirmation under the form of HK data packet. Just as with any other HK data packet, the TC echo packet must be requested by the operator and send over the HK feed from the Central Control Computer in order to be reviewed.

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